

Find the residual stresses at the top, mid, and bottom surfaces of the 90° ply in a [0/90]_s carbon/epoxy laminate subjected to a temperature change of -100°C. Each lamina is 0.127 mm thick. Use the following properties of the unidirectional carbon/epoxy lamina:

$$[Q] = \begin{bmatrix} 141 & 3 & 0 \\ 3 & 10 & 0 \\ 0 & 0 & 5 \end{bmatrix} \text{ GPa}; \alpha_1 = -0.9 \times 10^{-6} / ^\circ\text{C}; \alpha_2 = 27 \times 10^{-6} / ^\circ\text{C} \text{ and total strain is zero.}$$

$$\Delta T = -100^\circ\text{C}$$

$$\begin{bmatrix} \sigma_x^P \\ \sigma_y^P \\ \tau_{xy}^P \end{bmatrix} = \begin{bmatrix} \bar{Q}_{11} & \bar{Q}_{12} & \bar{Q}_{16} \\ \bar{Q}_{12} & \bar{Q}_{22} & \bar{Q}_{26} \\ \bar{Q}_{16} & \bar{Q}_{26} & \bar{Q}_{66} \end{bmatrix} \left\{ \begin{bmatrix} \epsilon_x^* \\ \epsilon_y^* \\ \gamma_{xy}^* \end{bmatrix} + z \begin{bmatrix} k_x \\ k_y \\ k_{xy} \end{bmatrix} - \begin{bmatrix} \epsilon_x^* \\ \epsilon_y^* \\ \gamma_{xy}^* \end{bmatrix} \right\}$$

$$= - \begin{bmatrix} \bar{Q}_{11} & \bar{Q}_{12} & \bar{Q}_{16} \\ \bar{Q}_{12} & \bar{Q}_{22} & \bar{Q}_{26} \\ \bar{Q}_{16} & \bar{Q}_{26} & \bar{Q}_{66} \end{bmatrix} \begin{bmatrix} \epsilon_x^* \\ \epsilon_y^* \\ \gamma_{xy}^* \end{bmatrix} \quad \text{--- (1)}$$

$$\begin{bmatrix} \epsilon_x^* \\ \epsilon_y^* \\ \gamma_{xy}^* \end{bmatrix} = \begin{bmatrix} c & s & -cs \\ s & c & cs \\ 2cs & -2cs & c^2 - s^2 \end{bmatrix} \begin{bmatrix} \epsilon_1^* \\ \epsilon_2^* \\ 0 \end{bmatrix};$$

$$= \begin{bmatrix} 0 & 1 & 0 \\ 1 & 0 & 0 \\ 0 & 0 & -1 \end{bmatrix} \begin{bmatrix} 0.9 \times 10^{-4} \\ -27 \times 10^{-4} \\ 0 \end{bmatrix}$$

$$= \begin{bmatrix} -27 \times 10^{-4} \\ 0.9 \times 10^{-4} \\ 0 \end{bmatrix} \text{ m/m}$$

$$\begin{bmatrix} \epsilon_1^* \\ \epsilon_2^* \\ \gamma_{12}^* \end{bmatrix} = \Delta T \begin{bmatrix} \alpha_1 \\ \alpha_2 \\ 0 \end{bmatrix} = -100 \begin{bmatrix} -0.9 \times 10^{-6} \\ 27 \times 10^{-6} \\ 0 \end{bmatrix} = \begin{bmatrix} 0.9 \times 10^{-4} \\ -27 \times 10^{-4} \\ 0 \end{bmatrix}$$

$$\begin{bmatrix} \bar{Q}_{11} & \bar{Q}_{12} & \bar{Q}_{16} \\ \bar{Q}_{12} & \bar{Q}_{22} & \bar{Q}_{26} \\ \bar{Q}_{16} & \bar{Q}_{26} & \bar{Q}_{66} \end{bmatrix}_{90} = \begin{bmatrix} 10 & 3 & 0 \\ 3 & 141 & 0 \\ 0 & 0 & 5 \end{bmatrix}$$

$$\begin{bmatrix} \sigma_x^P \\ \sigma_y^P \\ \tau_{xy}^P \end{bmatrix} = \begin{bmatrix} 10 & 3 & 0 \\ 3 & 141 & 0 \\ 0 & 0 & 5 \end{bmatrix} \begin{bmatrix} -27 \times 10^{-4} \\ 0.9 \times 10^{-4} \\ 0 \end{bmatrix} = \begin{bmatrix} -0.02673 \\ 4.59 \times 10^{-3} \\ 0 \end{bmatrix} \text{ GPa}$$

Ans: